

SUPPLEMENTAL TYPE CERTIFICATE

10087126

This Certificate/Approval is issued by EASA, acting in accordance with Regulation (EU) 2018/1139 on behalf of the European Union, its Member States and of the European third countries that participate in the activities of EASA under Article 129 of that Regulation and in accordance with Commission Regulation (EU) No. 748/2012 to

ALPINE AEROTECH LP

**1260 INDUSTRIAL ROAD
WEST KELOWNA BC V1Z 1G5
CANADA**

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and, if applicable, environmental protection requirements when operated within the conditions and limitations specified below:

Type Certificate Number: EASA.IM.R.106

Type Certificate Holder: BELL TEXTRON, INC.

Type: Bell 212/412

Model: 212

412

412EP

Original STC Number: TCCA SH21-26

Description of Design Change:

Center Pedestal Riser / The Center Pedestal Riser increases the height of the pedestal at the aft end while the height at the forward end is unchanged.

EASA Certification Basis:

The Type and OSD Certification Bases for the original product remains applicable to this certificate/ approval. The requirements for environmental protection and the associated certified noise and/ or emissions levels of the product are unchanged and remain applicable to this certificate/ approval without any impact on the noise database.

Associated Technical Documentation:

-Alpine Aerotech Lp Master Data List AAL-280-094-001 Rev. E dated 2023-12-01.

See Continuation Sheet(s)

For the European Union Aviation Safety Agency

Cologne, Germany, 19 May 2025



Marco Dioli
Section Manager Heavy Rotorcraft



-Infinion Master Data List MDL-2104-01 Issue 1 dated 9 April 2021.
-Installation Instruction AAL-280-095-701, Revision NI dated March 5, 2021.
-Instructions for Continued Airworthiness AAL-214-035-701 Rev. B dated 2023-12-01.
Or later revisions of the above listed document(s) approved/accepted on behalf of EASA in accordance with the Technical Implementation Procedures of EU/ Canada Bilateral Agreement.

Limitations/Conditions:

Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/ or repair will introduce no adverse effect upon the airworthiness of the product.

- End -