

SUPPLEMENTAL TYPE CERTIFICATE

10060733

This Supplemental Type Certificate is issued by EASA, acting in accordance with Regulation (EC) No. 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation and in accordance with Commission Regulation (EU) No. 748/2012 to:

ALPINE AEROTECH LP

1260 INDUSTRIAL ROAD
WEST KELOWNA BC V1Z 1G5
CANADA

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and environmental protection requirements when operated within the conditions and limitations specified below:

Original Type Certificate Number: 1. EASA.IM.R.106
2. US H1SW

Type Certificate Holder: BELL HELICOPTER TEXTRON, INC.

Type: 1. Bell 212/412
2. Bell 204/205/210

Model: 1. 212, 412, 412EP
2. 205A-1

Original STC Number: TCCA SH16-36 ISSUE 2

Description of Design Change:

Installation of Baggage Compartment Protection Kit defined with Alpine Aerotech Master Data List AAL-221-054-001.

See Continuation Sheet(s)

For the European Aviation Safety Agency

Date of Issue: 18 January 2017



Pier Giorgio COLOMBO
Medium Rotorcraft Section
Manager

10046133
SUPPLEMENTAL TYPE CERTIFICATE - 10060733 - ALPINE AEROTECH LP - 300552

EASA Certification Basis:

The Certification Basis (CB) for the original product remains applicable to this certificate/ approval.
The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.

Associated Technical Documentation:

- Alpine Aerotech Master Data List Document Number AAL-221-054-001 Revision D, dated 26 August 2016.
- Alpine Aerotech Installation Instructions Document Number AAL-221-055-005 Revision NI, dated 11 July 2016.
- Alpine Aerotech Instructions for Continuing Airworthiness Document Number AAL-221-055-701 Revision D, dated 26 August 2016.

Limitations/Conditions:

Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.

- End -