

## SUPPLEMENTAL TYPE CERTIFICATE

**10053796 REV. 1**

This Supplemental Type Certificate is issued by EASA, acting in accordance with Regulation (EC) No. 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation and in accordance with Commission Regulation (EU) No. 748/2012 to:

### ALPINE AEROTECH LP

1260 INDUSTRIAL ROAD  
WEST KELOWNA BC V1Z 1G5  
CANADA

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and environmental protection requirements when operated within the conditions and limitations specified below:

**Original Type Certificate Number:** US H4SW

**Type Certificate Holder:** BELL HELICOPTER TEXTRON, INC.

**Type:** BELL 212/412

**Model:** 212, 412, 412EP

**Original STC Number:** TCCA SH96-82 ISSUE 2

**Description of Design Change:**

Alpine Aerotech Ltd. Stiffener Assembly

Installation of stiffeners to the firewall assembly in accordance with Alpine Aerotech, Ltd. Master Document List 191M-96-028-MDL Rev.NI.

**EASA Certification Basis:**

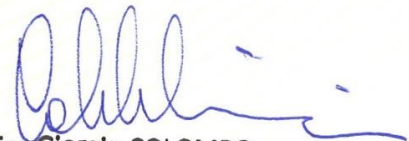
The Certification Basis (CB) for the original product remains applicable to this certificate/ approval.

The requirements for environmental protection and the associated certified noise and/ or emissions levels of

See Continuation Sheet(s)

**For the European Aviation Safety Agency**

**Date of Issue:** 06 June 2017



**Pier Giorgio COLOMBO**  
**Medium Rotorcraft Section**  
**Manager**

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the original product are unchanged and remain applicable to this certificate/ approval.

**Associated Technical Documentation:**

Alpine Aerotech Ltd. Master Document List 191M-96-028-MDL Rev.NI

Alpine Aerotech Ltd Illustrated Part Breakdown 191M-96-028-IPB Rev. NI

Alpine Aerotech Ltd. Installation Instructions Document No. 191M-96-028-II, Rev.B

Alpine Aerotech Ltd. Maintenance Manual Document No. 191M-96-028-MMS, Rev.NI.

**Limitations/Conditions:**

Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.

- End -

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