

SUPPLEMENTAL TYPE CERTIFICATE

10062577 REV. 1

This Supplemental Type Certificate is issued by EASA, acting in accordance with Regulation (EC) No. 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation and in accordance with Commission Regulation (EU) No. 748/2012 to:

ALPINE AEROTECH LP

1260 INDUSTRIAL ROAD
WEST KELOWNA BC V1Z 1G5
CANADA

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and environmental protection requirements when operated within the conditions and limitations specified below:

Original Type Certificate Number: EASA.IM.R.106

Type Certificate Holder: BELL HELICOPTER TEXTRON, INC.

Type: Bell 212/412

Model: 212/ 412/ 412EP

Original STC Number: TCCA SH13-13 ISSUE 1

Description of Design Change:

Aft Cabin Wall Protection Kit (TCCA STC SH13-13, Issue 1).

Installation of polymer panels on top of the cabin pylon and aft cabin composite panel walls, to provide protection from damages and wear. Polymer panels are bolted on inserts already installed on the native composite panels.

EASA Certification Basis:

The Certification Basis (CB) for the original product remains applicable to this certificate/ approval.

See Continuation Sheet(s)

For the European Aviation Safety Agency

Date of Issue: 25 July 2017



Pier Giorgio COLOMBO
Medium Rotorcraft Section
Manager

10050244

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The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.

Associated Technical Documentation:

AAL-212-030-701, Revision A, dated 2015-09-29.

AFT CABIN BULKHEAD PROTECTOR, KIT INSTRUCTIONS FOR CONTINUING AIRWORTHINESS or later approved revisions from TCCA.

Doc. No. AAL-212-040-701, Revision B, dated 2015-10-26.

PYLON PANEL PROTECTOR, KIT INSTRUCTIONS FOR CONTINUING AIRWORTHINESS or later approved revisions from TCCA.

Limitations/Conditions:

Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/ or repair will introduce no adverse effect upon the airworthiness of the product.

- End -